

SUPPLEMENTAL TYPE CERTIFICATE

10060937

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

ANJOU AERONAUTIQUE S.R.L.

STR. LIVEZII 98, ETAJ P 550042 SIBIU **ROMANIA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.A.015

Type Certificate Holder: AIRBUS S.A.S.

Type: A340

Model: A340-312/-313

Description of Design Change:

A340-300 SEAT BELTS MODIFICATION

The design change related to this STC consist in the replacement of existing safety belts by new ones manufactured by ANJOU AERONAU. The modification is on BC seats type 1404 & EC seats type 9174 and 3509.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Munda

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 07 February 2017

Colin HANCOCK Section Manager **Supplemental Type Certificates (STCs)** & Special Projects

10042629

SUPPLEMENTAL TYPE CERTIFICATE - 10060937 - ANJOU AERONAUTIQUE S.R.L. - 306082





Associated Technical Documentation:

- Certification Plan ref 02-610850-CP-001 Rev 04
- Master Data List ref 02-610850-MDL-001 Rev 02
- ICA: CMMS (Component Maintenance Manual Supplement), ref 02-610850-CMMS-003 Rev 1
- Modification Approval Sheet, Project 02-610850 Rev 02
- or later revisions of the above listed documents approved by EASA.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

